THE USE OF MATHEMATICS IN INTENSITY ESTIMATION WIND FOR UNMANNED AIR VEHICLES

Dorin AFANAS, doctor, associate professor https://orcid.org/0000-0001-7758-943X Department of Algebra, Geometry and Topology

Abstract. There are various weather risks that the operators of unmanned aerial vehicles must take into account and which depend on the following properties of the atmosphere $[1, p. 82]$: • Altitude density; • Temperature; • Visibility; • Humidity and saturation; • The wind. In this article we will investigate the estimation of the wind speed at the requested height.

Keywords: unmanned aerial vehicle, wind intensity, height, anemometer, Beaufort scale, functional dependence, formula.

Rezumat. Există diverse riscuri meteo de care operatorii vehiculelor aeriene fără pilot trebuie să țină cont și care depind de următoarele proprietăți ale atmosferei [1, p. 82]: • Densitatea altitudinii; • Temperatura; • Vizibilitatea: • Umiditatea și saturația: • Vântul. În prezentul articol vom cerceta estimarea vitezei vântului la înălțimea solicitată.

Cuvinte cheie: vehicul aerian fără pilot, intensitatea vântului, înăltime, anemometru, scara lui Beaufort, dependență funcțională, formulă.

Due to the small size, during the flight of an unmanned aerial vehicle, an important factor in the stability of its handling is the environmental conditions. One of these factors is the intensity equivalent to the wind speed at the height where this flight will take place. That is why it is very important that before the start of the flight we can estimate the intensity of the wind at that height.

The device with which we can determine the wind speed is called an anemometer (fig. 1). To be able to estimate the wind speed at the required height, the formula is usually used:

$$
V_1 = V_0 \cdot \left(\frac{H_1}{H_0}\right)^k
$$

where V_1 is the wind speed at the requested height;

 V_0 – wind speed at the measured height;

 H_0 – the height at which the measurements take place (for weather stations, $H_0 = 10$ m is considered);

 H_1 – requested height;

 k is an empirical coefficient and can take one of the following values shown in the table below:

Figure 1. Anemometer

Students should also be trained in visual assessment of wind intensity equivalent to speed at a height of 10 m .

For this purpose, Beaufort's scale is presented:

After the presentation of the above concepts, laboratory works are carried out, which elucidate the functional dependence $V_1 = f(V_0, H_0, H_1)$.

Below are various type dependencies $V_1 = f(V_0, H_0, H_1)$ for completely open terrain such as airport runways.

Estimated wind intensity equivalent to speed at 30 m height, based on speed measured at $2m$ height, for fully open terrain such as airport runways:

Estimation of wind intensity equivalent to speed at heights between 10 m and 120 m , depending on speed $V_0 = 2$ m/s measured at a height of 2 m, for completely open ground such as airport runways:

Estimation of the wind intensity equivalent to the speed at heights between 10 m and 120 m, depending on the speed $V_0 = 4$ m/s measured at the height of 2 m, for a completely open terrain, such as airport runways:

In case we want to check the correctness of the formula:

$$
V_1 = V_0 \cdot \left(\frac{H_1}{H_0}\right)^k
$$

it is important that the measurements taken at the ground surface are taken at the same time as those at the requested height. For this purpose, a device that measures the wind intensity is placed on board the unmanned aerial vehicle, after which the vehicle is maintained at the respective height.

In the absence of such a device, one method would be to travel the same distance at a constant speed by the air vehicle against and in the direction of the wind timing the time, after which we can easily determine the required data. But this method does not always give us accurate data.

Another laboratory work can be done based on the displacement calculation formula:

$$
S=\frac{v_A}{nD'}
$$

where v_A is the velocity of the axial current, and nD – the speed of the outer points on the propeller circle.

This being the abscissa of the free movement diagram, we can express it through the functions:

- $K_T(S) = \frac{T}{\rho n^2 D^4}$ the value of Schube; $K_Q(S) = \frac{Q}{m^2 D^5}$ – the torque value;
- $\triangleright \quad \eta_o(S) = \frac{Tv_A}{2\pi n \omega} = \frac{S}{2\pi} \cdot \frac{K_T}{K_O}$ vane angle measure value.

Conclusions

A small unmanned aerial vehicle is not recommended for flight:

- when the wind is strong; \bullet
- when the speed of the gusts can exceed the maximum speed of the unmanned aerial \bullet vehicle:
- less than 40 km from a storm, because hail can fall a long way from storm clouds \bullet
- near tornadoes; \bullet
- during freezing rain, which are the most unfavorable weather conditions. This is \bullet explained by the fact that ice can be stored on the surface of the unmanned aerial vehicle (icing) and therefore can change the aerodynamic characteristics of the propellers;
- in foggy weather, because it is more difficult to maintain visual contact with the \bullet unmanned aerial vehicle.

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